

indexes of the rotor-supporting system, such as phase trajectory, power spectrum, the Poincare's mapping, and the overall bifurcation diagram. Section 3 discusses the appropriate control strategy to eliminate cycle times and chaotic motion based on the research results, so the active control on the bearing-rotor can be achieved. Finally, the conclusions and recommendations for further work are given.

II. MATHEMATICAL MODELING OF HSFD SYSTEM

Fig.1 is a principle diagram of HSFD and the rotor. It consists of a controllable hybrid bearing, a hydraulic half-bridge driven by piezoelectric actuator, a controller and a power supply. The rigid rotor is supported on the device by a roller bearing with HSFD and a retaining spring. A controllable damping force can be formed to achieve active damping by the squeeze-film damper. It divides the regular oil chamber into four independent static pressure regions at rotational direction. For the sake of convenience, the following assumptions of HSFD are made as

- A1: The oil in HSFD is incompressible fluid with the same viscosity;
- A2: The fluid between axis and dynamic pressure region is laminar flow;
- A3: The depth of oil chamber is much larger than the clearance of damper and the pressure in the oil chamber is considered to be a constant;
- A4: The elastic deformation between the bearing and the bearing housing is negligible and flow inertia is negligible;
- A5: The configuration of HSFD is symmetric and the size of each oil chamber is identical.

A. Pressure distribution equations

Based on the assumption of incompressible and oil film bearing theory in the reference [4], the Reynolds equation is given as follows

$$\frac{1}{R^2} \frac{\partial}{\partial \theta} \left(h^3 \frac{\partial p}{\partial \theta} \right) + \frac{\partial}{\partial z} \left(h^3 \frac{\partial p}{\partial z} \right) = 12\mu(\dot{\varepsilon} \cos \theta + \varepsilon \dot{\phi}_B \sin \theta) C \quad (1)$$

where R represents radius of oil film. $\theta = \phi - \phi_B$, $\varepsilon = e/C$ mean dimensionless eccentricity, C means clearance of radius, ϕ_B means the motion angle of bearing axis, μ means dynamic viscosity of lubricating oil, and p represents the pressure of oil film.

The pressure distribution of HSFD can be divided into 3 parts: static pressure region, rotation direction dynamic pressure region and axial direction dynamic pressure region. The pressure inside the static pressure range is constant. Each pressure $p_{c,i}$ of the four ranges is as follows:

$$\begin{cases} p_{c,1} = p_s - \Delta p_1, p_{c,2} = p_s - \Delta p_2 \\ p_{c,3} = p_s + \Delta p_1, p_{c,4} = p_s + \Delta p_2 \end{cases} \quad (2)$$

where p_s represents the pressure of oil supply. The differential pressure Δp_1 of range 1 and 3 and the differential pressure Δp_2 of range 2 and 4 are decided by the following equations.

k_p and k_d respectively on behalf of proportion and differential coefficients.

$$\begin{cases} \Delta p_1 = k_p x + k_d \dot{x} \\ \Delta p_2 = k_p y + k_d \dot{y} \end{cases} \quad (3)$$

In the polar system, Eq.(3) can be transformed as follows

$$\begin{cases} \Delta p_1 = k_p \varepsilon \cos \phi_B + k_d (\dot{\varepsilon} \cos \phi_B - \varepsilon \dot{\phi}_B \sin \phi_B) \\ \Delta p_2 = k_p \varepsilon \sin \phi_B + k_d (\dot{\varepsilon} \sin \phi_B + \varepsilon \dot{\phi}_B \cos \phi_B) \end{cases} \quad (4)$$

The pressure distributing of dynamic pressure range and axis can be decided based on solving Reynolds equations (1) after deciding each range's pressure as the boundary conditions.

In the area $-a \leq z \leq a$ of the HSFD, the pressure in the static pressure region is constant as $p_{c,i}$ and the pressure in the dynamic pressure region can be solved by using the long bearing theory. In the area $a \leq |z| \leq L/2$ of the HSFD, the pressure can be solved by using short bearing theory and the boundary condition of $p(z, \theta)|_{z=\pm a} = p_0(\theta)$ and $p(z, \theta)|_{z=\pm L/2} = 0$. The pressure distribution is shown in Fig.2 and Fig.3. and $p_0(\theta)$ can be written as

$$p_0(\theta) = \begin{cases} p_{c,i} & \frac{\pi}{2}(i-1) - \frac{\beta}{2} - \phi_B \leq \theta \leq \frac{\pi}{2}(i-1) + \frac{\beta}{2} - \phi_B \\ p_i(\theta) & \frac{\pi}{2}(i-1) + \frac{\beta}{2} - \phi_B \leq \theta \leq \frac{\pi}{2}i - \frac{\beta}{2} - \phi_B \end{cases} \quad (5)$$

According the pressure distribution assumed in Fig.3, solve the Eq.(1), we can get the pressure distribution $p_i(\theta)$ as follows

$$p_i(\theta) = p_{c,i} + \frac{6\mu R^2 \dot{\varepsilon}}{C^2 \varepsilon} \left[\frac{1}{(1+\varepsilon \cos \theta)^2} - \frac{1}{(1+\varepsilon \cos \theta_{i1})^2} \right] + C_1 \int_{\theta_{i1}}^{\theta} \frac{1}{C^3 (1+\varepsilon \cos \theta)^3} d\theta - \int_{\theta_{i2}}^{\theta} \frac{12\phi_B \mu C \varepsilon R^2 \cos \theta}{C^3 (1+\varepsilon \cos \theta)^3} d\theta \quad (6)$$

where $i=1,2,3,4$, $\theta_{i1} = (i-1)\frac{\pi}{2} + \frac{\beta}{2} - \phi_B$, and β is half of the distribution angle of static pressure region. Substituting $\theta = \theta_{i2} = i\frac{\pi}{2} - \frac{\beta}{2} - \phi_B$ into Eq.(6) yields the coefficient C_1 expression as follows

$$C_1 = \frac{p_{c,i+1} - p_{c,i} - \frac{6\mu R^2 \dot{\varepsilon}}{C^2 \varepsilon} C_2}{\int_{\theta_{i1}}^{\theta_{i2}} \frac{1}{C^3 (1+\varepsilon \cos \theta)^3} d\theta}, C_3 = \int_{\theta_{i1}}^{\theta_{i2}} \frac{12\phi_B \mu C \varepsilon R^2 \cos \theta}{C^3 (1+\varepsilon \cos \theta)^3} d\theta, C_2 = \frac{1}{(1+\varepsilon \cos \theta_{i2})^2} - \frac{1}{(1+\varepsilon \cos \theta_{i1})^2} + C_3.$$

In area of HSFD with $a \leq |z| \leq L/2$, the pressure distribution can be solved as follows

$$p(z, \theta) = \left(\frac{L}{2} - |z| \right) \left[[A_1(\theta) \dot{\phi}_B \varepsilon + A_2(\theta) \dot{\varepsilon}] (a - |z|) + p_0(\theta) / \left(\frac{L}{2} - a \right) \right] \quad (7)$$

where $A_1(\theta) = \frac{6\mu C \sin \theta}{C^3(1 + \varepsilon \cos \theta)^3}$, $A_2(\theta) = \frac{6\mu C \cos \theta}{C^3(1 + \varepsilon \cos \theta)^3}$.

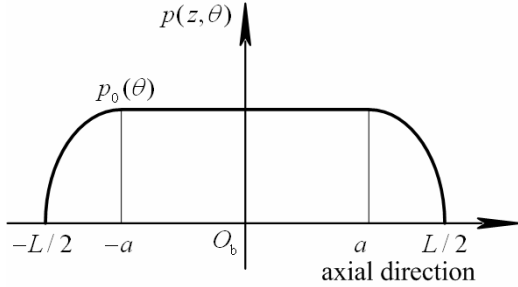


Figure 2. Pressure distribution of axial direction in HSF

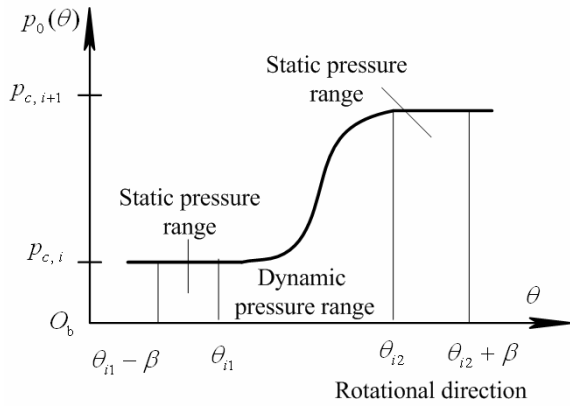


Figure 3. Pressure distribution of rotational direction in HSF

B. Solution of the instant oil film supporting force

The oil film forces of HSF oil pressure can be determined when the oil film pressure of static pressure region decided by Renault equations is achieved. The oil film force of HSF consists of 3 parts: static pressure oil film forces, axis direction dynamic pressure region forces and the rotational direction dynamics pressure region forces. The force components in the area of static pressure region and the dynamic pressure region can be respectively given as follows

$$\begin{cases} F_{sr} = 2aR \sum_{i=1}^4 (\sin \theta_{i1} - \sin(\theta_{i1} - \beta - \phi_B)) p_{c,i} \\ F_{st} = 2aR \sum_{i=1}^4 (\cos \theta_{i1} - \cos(\theta_{i1} - \beta - \phi_B)) p_{c,i} \end{cases} \quad (8)$$

$$\begin{cases} F_{tr} = \sum_{i=1}^4 \int_{\theta_{i1}}^{\theta_{i2}} p_i(\theta) \cdot 2aR \cos \theta d\theta \\ F_{tt} = \sum_{i=1}^4 \int_{\theta_{i1}}^{\theta_{i2}} p_i(\theta) \cdot 2aR \sin \theta d\theta \end{cases} \quad (9)$$

where $p_i(\theta)$ can be determined by Eq.(6). In the area of $a \leq |z| \leq L/2$ at axis direction, with the symmetry character, the oil film forces supplied by the oil film pressure forces can be determined by the equations as follows

$$\begin{cases} F_{ar} = 2 \int_a^{L/2} dz \int_0^{2\pi} p(z, \theta) R \cos \theta d\theta \\ F_{at} = 2 \int_a^{L/2} dz \int_0^{2\pi} p(z, \theta) R \sin \theta d\theta \end{cases} \quad (10)$$

Based on the above-analyzed results, the total oil film forces can be solved as $F_r = F_{sr} + F_{tr} + F_{ar}$, and $F_t = F_{st} + F_{tt} + F_{at}$.

C. Rotor Dynamics Equation

The force diagram of rotor support system shows in Fig.4, O_b is center of oil film ring, O_j is oil film journal center, O_c is gravity center. The load can be got through the motion and force analysis: unbalanced exciting force caused by mass eccentricity, centrifugal inertia force brought by motion, oil film force and elastic restore force of bearing, and inertia force in journal center created by precession acceleration that around the oil film ring.

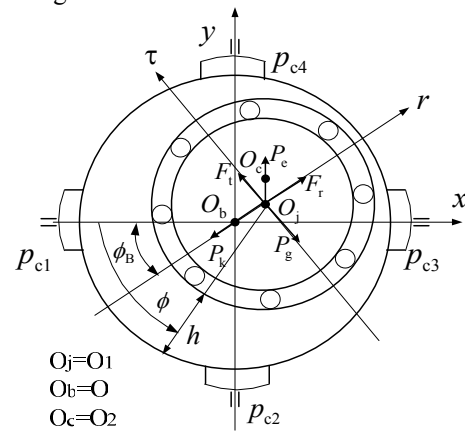


Figure 4. Force analysis of a rotor with HSF

Assuming that the rotor speed is ω_s , v is mass relative eccentricity, $e_m = U_c$ is mass eccentricity, m is the rotor mass that lumped at the mid-point. By using free-body method to carry out the analysis to the action forces applied to rotor-supporting system, the dynamical model is described as follows

$$\begin{cases} \ddot{\varepsilon} - \varepsilon \dot{\phi}_B^2 = h_1 + v \cos(\omega_s t - \phi_B) - h_2 \varepsilon - 2\xi \dot{\varepsilon} / \omega \\ \varepsilon \ddot{\phi}_B + 2\dot{\varepsilon} \dot{\phi}_B = h_3 + v \sin(\omega_s t - \phi_B) - 2\xi \dot{\phi}_B / \omega \end{cases} \quad (11)$$

where $\omega = \frac{\omega_s}{\omega_c}$, $h_1 = \frac{F_r}{mC\omega_s^2}$, $h_2 = \frac{K}{mC\omega_s^2}$, $h_3 = F_t / (mC\omega_s^2)$.

Combining Eq.(11) with Eq.(2) to Eq.(10), yields the mathematical model of the overall HSF system.

III. DYNAMIC DIGITAL SIMULATION OF HSFD

A. Procedure of the digital simulation

Fig.5 shows the whole computing simulation process of HSFD dynamic characteristics. In the start of first calculation, a set of initial values $\varepsilon_0, \dot{\varepsilon}_0, \phi_{B0}, \dot{\phi}_{B0}$ are taken into PD control Eq.(4) to confirm the initial pressure $p_{c,i}$ in static pressure chamber, and then the oil film pressures of every position can be solved through taking $p_{c,i}$ into Eq.(5) and Eq.(7). With that the oil film forces in static pressure chamber, rotational pressure range and axial pressure range are achieved in Eq.(8-10). Afterward, taking the above three forces into Eq.(11) and the $\varepsilon, \dot{\varepsilon}, \phi_B, \dot{\phi}_B$ could be got by solving second-order differential equations with Runge-Kutta method, then a new turn iteration process is started by taking the first solutions of $\varepsilon, \dot{\varepsilon}, \phi_B, \dot{\phi}_B$ into Eq.(4).

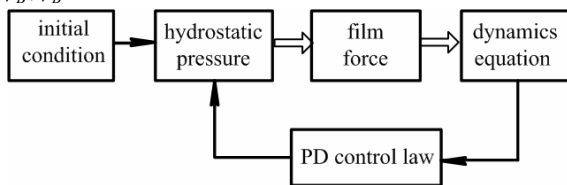


Figure 5. Calculating procedure diagram

The main parameters are as follows: damping ratio $\xi = 0.0005$, oil chamber numbers $n=4$, critical speed $\omega_c = 140\pi$ rad/s, oil density $\rho = 850$ kg/m³, $\mu = 0.008$ m²/s, $a=0.012$ m, $L=0.04$ m, $R=0.03$ m, $p_s = 1 \times 10^6$ Pa, $C=5.3 \times 10^{-4}$ m.

Runge-Kutta algorithm is used to solve the dynamics equations. On step division, the computing experience in reference [3] will be consulted, that means 540 points will be picked out in rotor one cycle rotation and the instant data of the first cycles are abandoned in order to ensure steady state motion can be arrived.

Define abbreviations and acronyms the first time they are used in the text, even after they have been defined in the abstract. Abbreviations such as IEEE, SI, MKS, CGS, sc, dc, and rms do not have to be defined. Do not use abbreviations in the title or heads unless they are unavoidable.

B. Parameter working scale discussion of HSFD active control program

The computing results above show the bifurcation and chaos dynamics characteristics as the rotating speed ratio ω is variation in the condition that the proportional control coefficient k_p is fixed; besides, the same characteristics are existed when ω is fixed and k_p is variation. So it is proved that the period-doubling and chaos motion can be completely eliminated to make the trace of rotor axial center always keep the state of limit loop as long as properly regulating coefficient k_p , when in the condition of fixed rotating speed. It is very significant to enhance rotor speed and increase the fatigue lifespan.

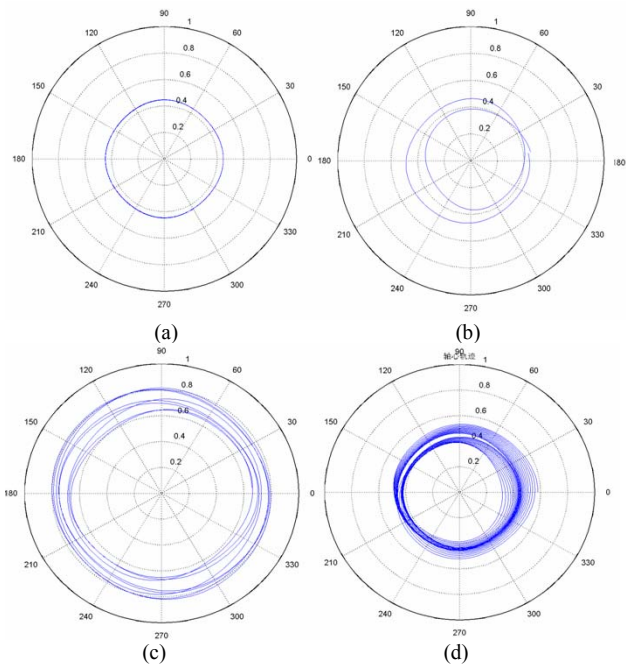


Figure 6. Relationship between the solution types and parameter k_p

The whole computing process adopts MATLAB language program and divides rotor's one rotation cycle into 540 steps to calculate. Taking $\omega=1.2$ for example, the simulation will choose the initial value $\varepsilon_0 = 0.6(X=0.6, Y=0)$ according to assumption of coordination circle precession, then to observe the solution through regulating the proportional control coefficient k_p . Fig.6 (a)-(d) are the motion traces of rotor axial center in condition of the k_p values to be 1.1,1.6,1.7,1.74 respectively. There into, Fig.6(b) and (c) are trajectories of 2-period solution and 4-period solution separately, and there may exists chaos solutions when on some ω values and k_p is in certain scale (e.g. $k_p=1.74$). So it means that keeping the rotor dynamics states always be 1-period solution as long as choosing proper k_p value scale in appropriate rotating speeds. For every ω value, a corresponding k_p interval which makes the rotor stay in the state of 1-period solution can be obtained, and Fig.7 shows the diagram. Rotor only stays in the state of 1-period solution no matter which speed rotor has, when k_p adopts values in the shadow area.

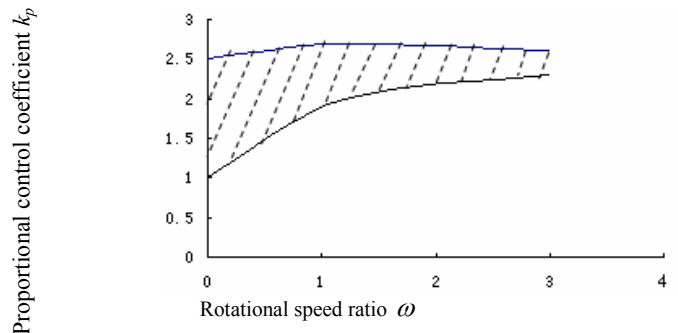


Figure 7. Proportional gain k_p versus Rotational speed ratio ω when the axis centre trajectory keeps 1-period solution

In the implementation process, rotor rotating speed ratio is easily measured by speed sensor; and it is convenient to adjust the proportional coefficient of PD controller for the computer control system. This procedure is actually to determine a series of $\omega \rightarrow k_p$ maps. If $\omega \rightarrow k_p$ map is known, the active control of high speed rotor can be implemented in the condition of choosing proper coefficient k_p according to the measured rotating speed ratio ω .

IV. CONCLUSIONS

In the paper, it is studied that rotor supporting system of squeeze-film damper with active control has two motion states of bifurcation and chaos by using the digital simulation for HSFD system. The complicated bifurcation and chaos dynamics characteristics of HSFD system have been analyzed by the quantitative and qualitative methods from the diagrams of phase traces and bifurcation charts.

Furthermore, it is found that HSFD system also shows the bifurcation and chaos dynamics states in some fixed speed, as the proportional coefficient k_p variation. So the motions of 2-period and chaos in system can be eliminated and to make HSFD system stay in 1-period state, as long as choosing the proper proportional differential controller. Based on this, an active control idea for high speed rotor in HSFD system was

put forward. It is significant to increase rotor working speed and fatigue lifespan and decrease body vibration.

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